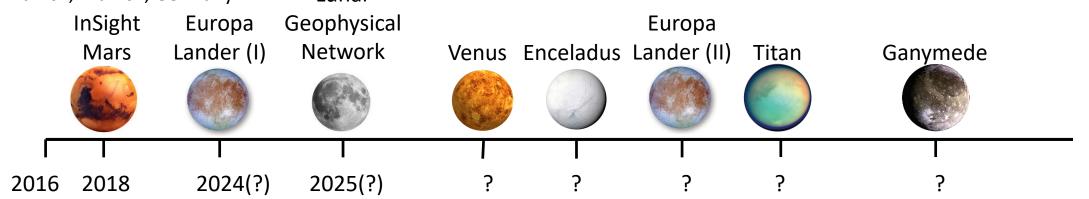


Geophysical Tests for Habitability in Europa and Other Ocean Worlds

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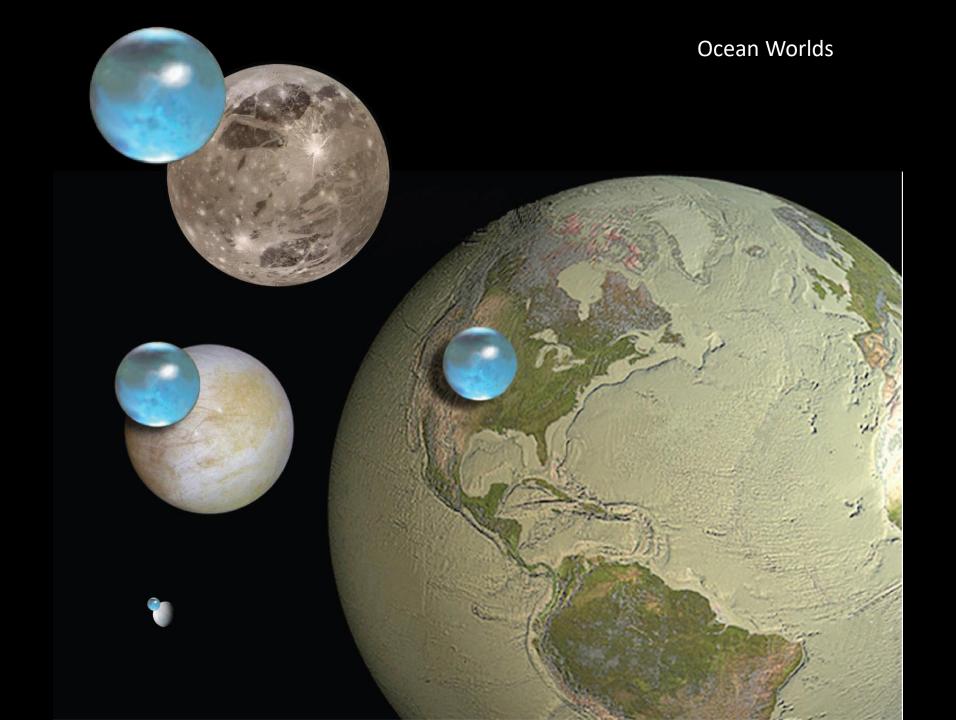
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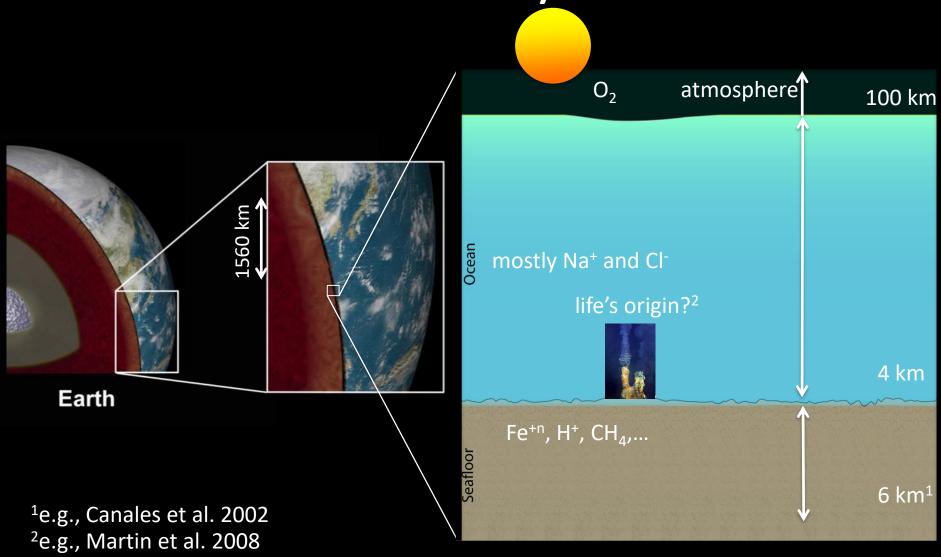
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Seismology in Ocean Worlds

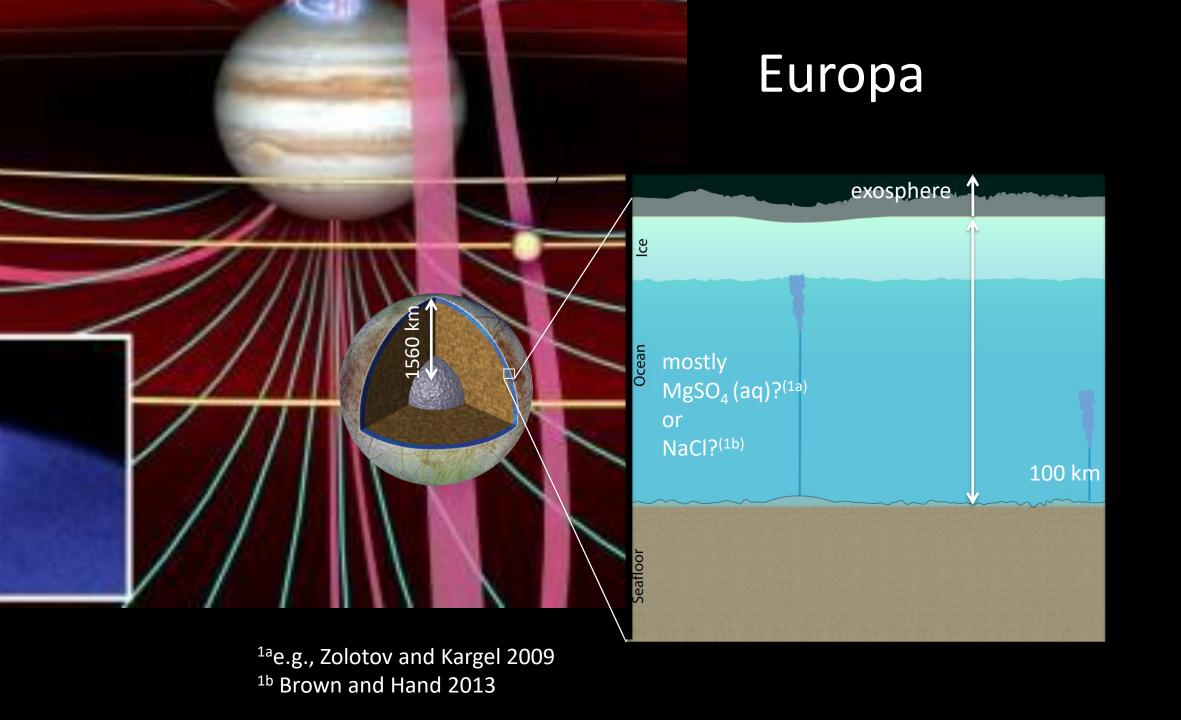
- How might ocean worlds be habitable?
 - Need to understand redox fluxes and reservoirs
- How might we use geophysics to test for habitability?
 - Focus here is on seismology

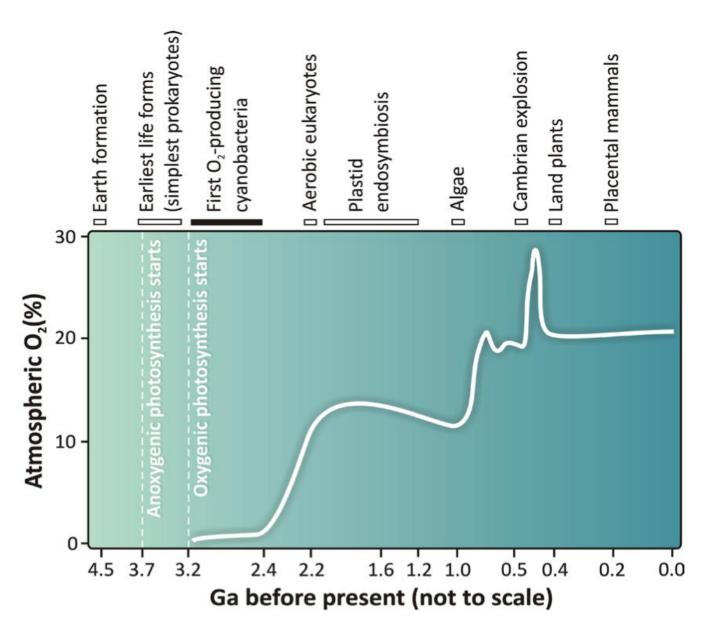


The Earth System



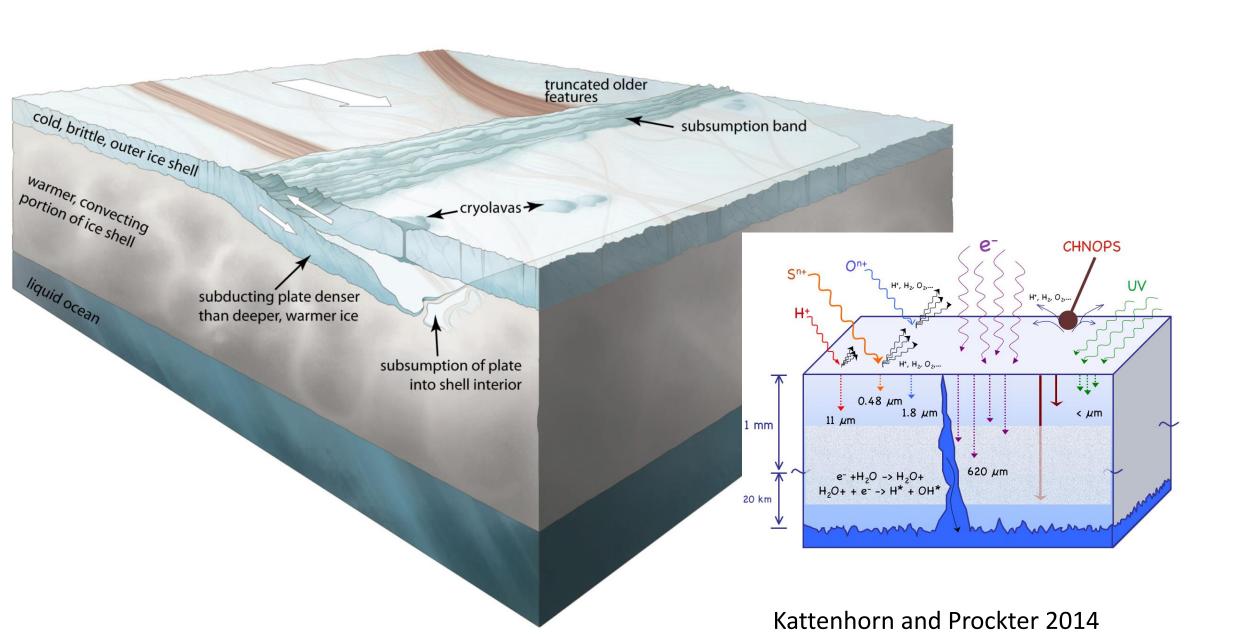
Earth Images: NASA

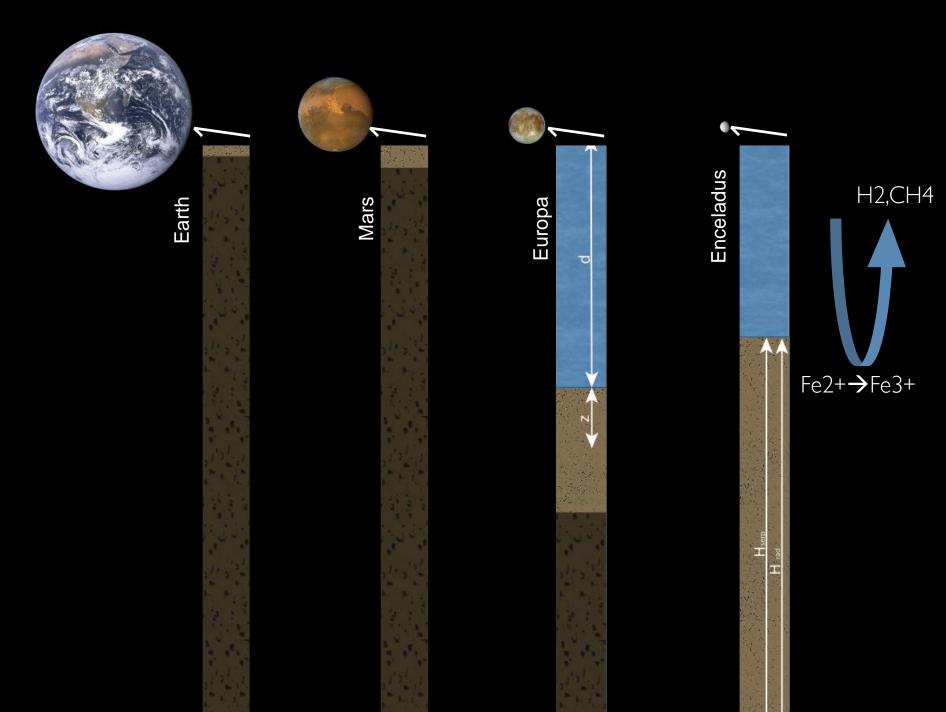




From "Adventures with cyanobacteria: a personal perspective," Govindjee et al. 2011

"Plate Tectonics" on Europa

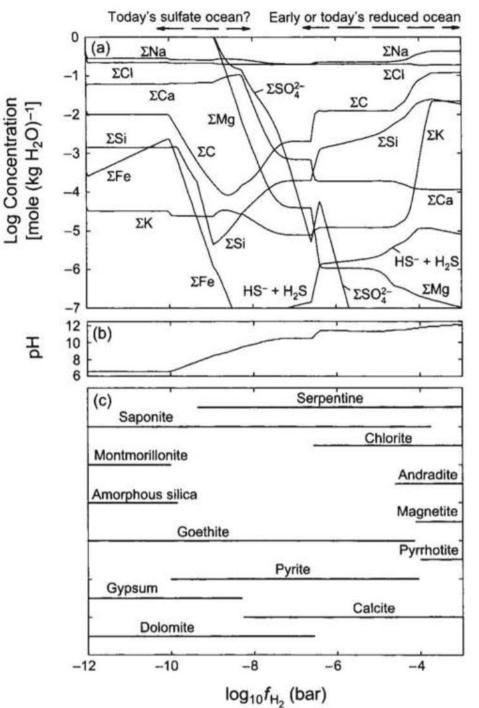






MgSO₄

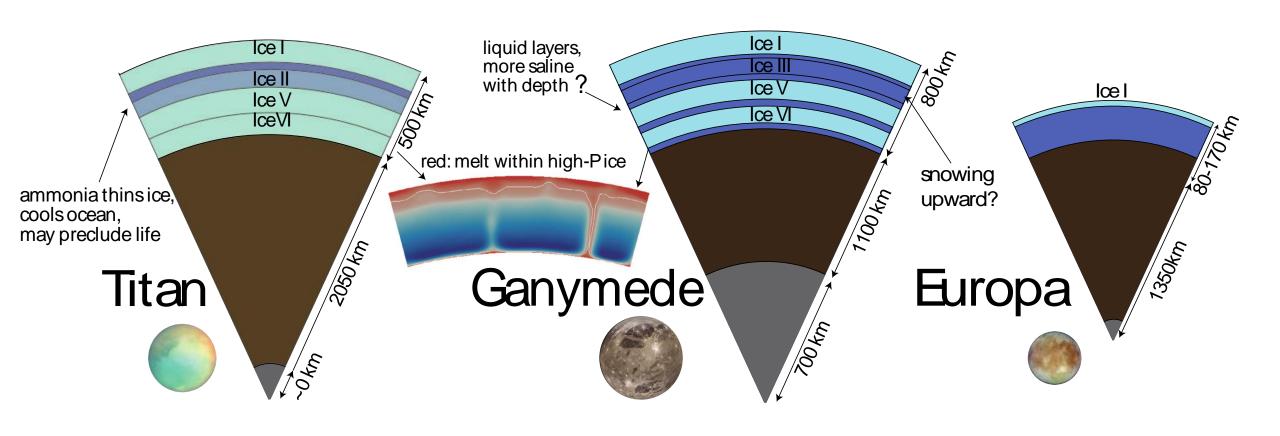






NaCl

Zolotov and Kargel 2009



Vance + 2014

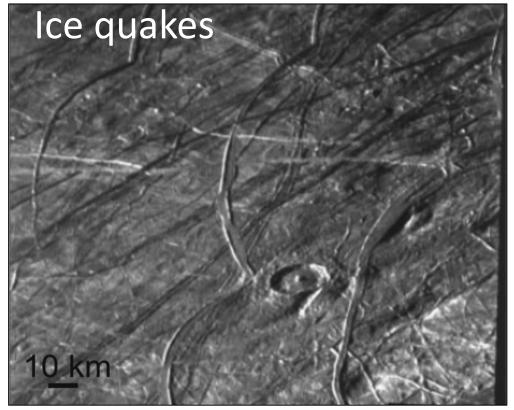
Choblet + 2017

11

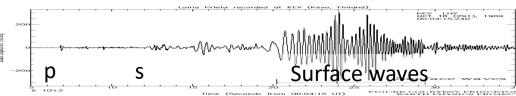
	Mean Radius in km	Bulk Density in kg m ⁻³	Moment of Inertia (C/MR ²)	Inferred H ₂ O thickness in km	Rotation Period In hrs	Radiogenic Heat in GW	Tidal Dissipation in GW	Seismic Energy Release in GW (%)
Earth ^a	6371	5514	0.3307	3.5	24	31,000	2636±16	22 (0.8)
Moon ^{a,b}	1738	3340	0.3929 ± 0.0009	n/a	672	420	1.36±0.19	6x10 ⁻⁷ (0.4x10 ⁻⁵)
Mars ^c	3397	3933	0.3662 ± 0.0017	n/a	24.6230	3,300		?
Europa ^{d,e,f}	1565.0±8.0	2989±46	0.346 ± 0.005	80-170	84.4	200	>1000 (ice) >1000 (ocean) 1-10 (rock)	?
Ganymede ^{e,f,g}	2631.17±1.7	1942.0±4.8	0.3115±0.002 8	750-900	171.7	500	10 ³ -10 ⁴ (ice) >1 (ocean)	?
Callisto ^{e,f}	2410.3±1.5	1834.4±3.4	0.3549±0.004 2	350-450	400.5	400	>1 (ice) >4 (ocean)	?
Titan ^{e,h,i}	2574.73 ±0.09	1879.8±0.2	0.3438±0.000 5	500-700	382.7	400	>60 (ice) >11 (ocean)	?
Enceladus ^{e,h,j}	252.1±0.1	1609±5	0.335	60-80	32.9	0.3	>20 >10 (ocean)	?

a) Williams et al., 2001 b) Goins et al., 1981; Williams et al., 1996; Siegler and Smekrar, 2014 c) Folkner et al., 1997, Nimmo and Faul, 2013 d) Tobie et al. 2003, Hussmann et al., 2006; Vance et al., 2007 e) Chen et al., 2014; Tyler, 2014 f) Schubert et al. 2004 g) Bland et al., 2015 h)

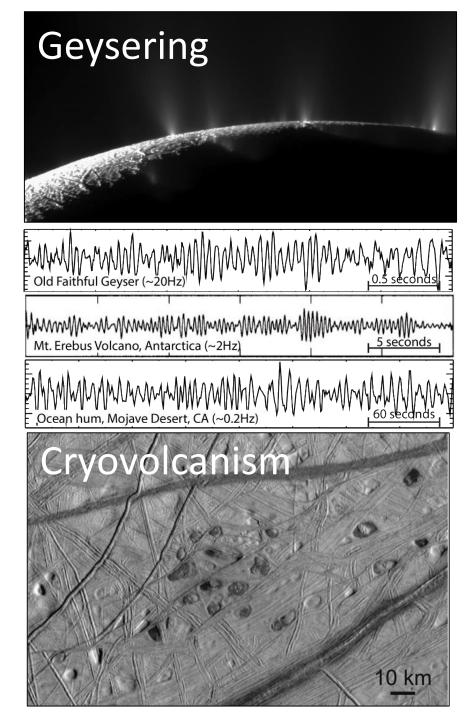
How might an ocean world sound?



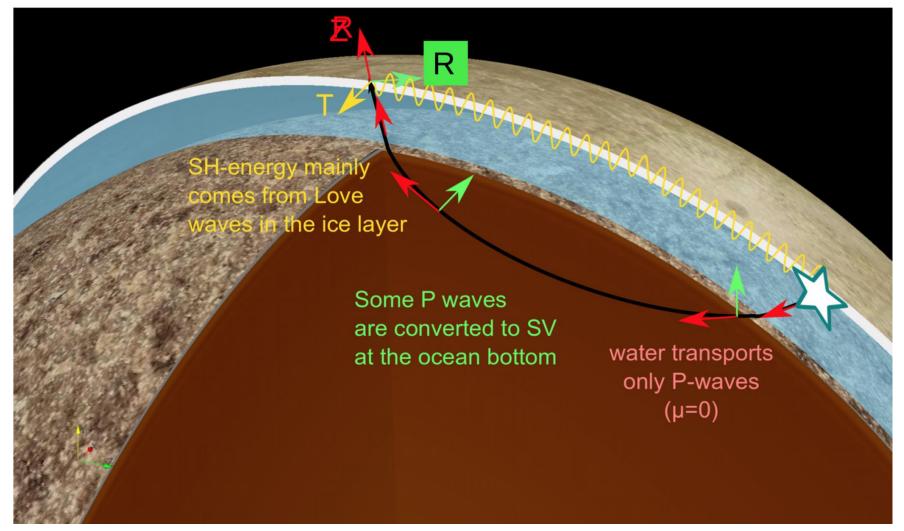
Fluid flow induced seismicity: continuous background "hum"



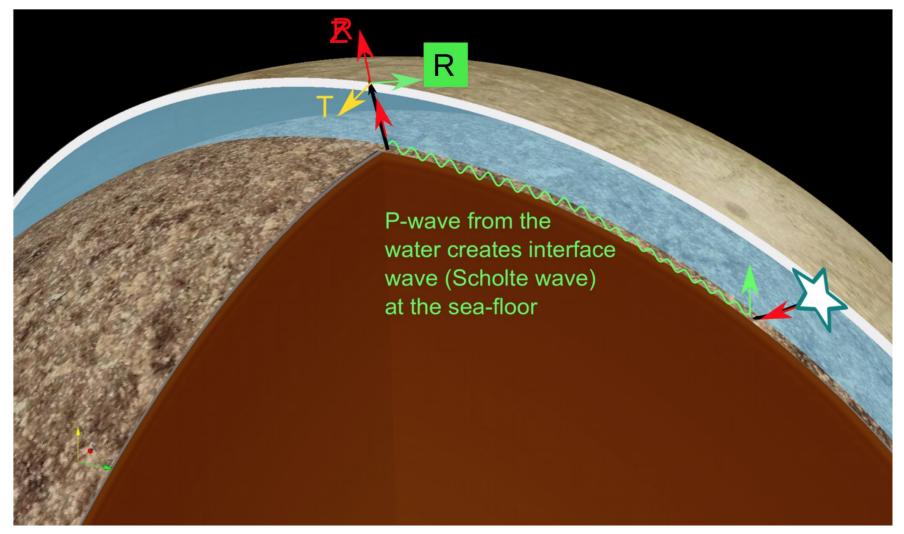
Europa's ice is young and geologically active



Sensing the mantle and ice

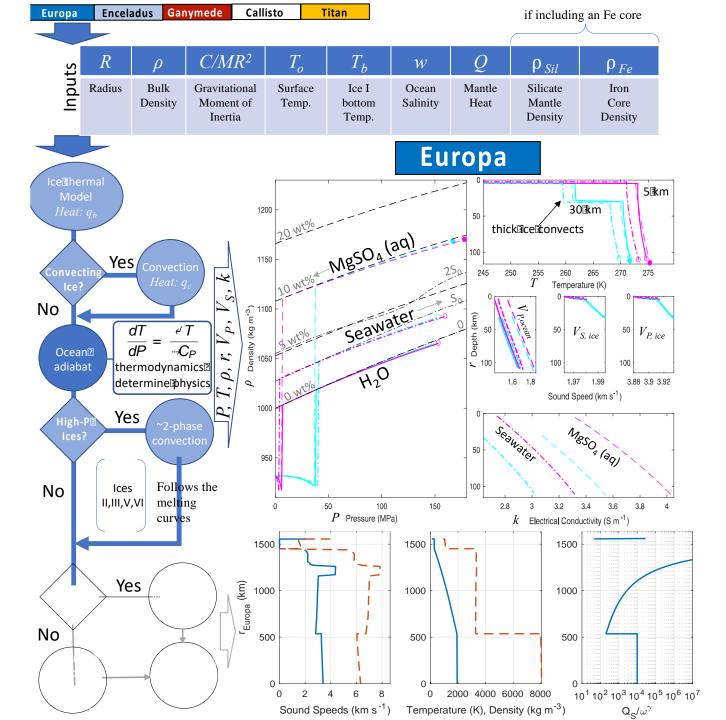


Sensing the mantle and ice

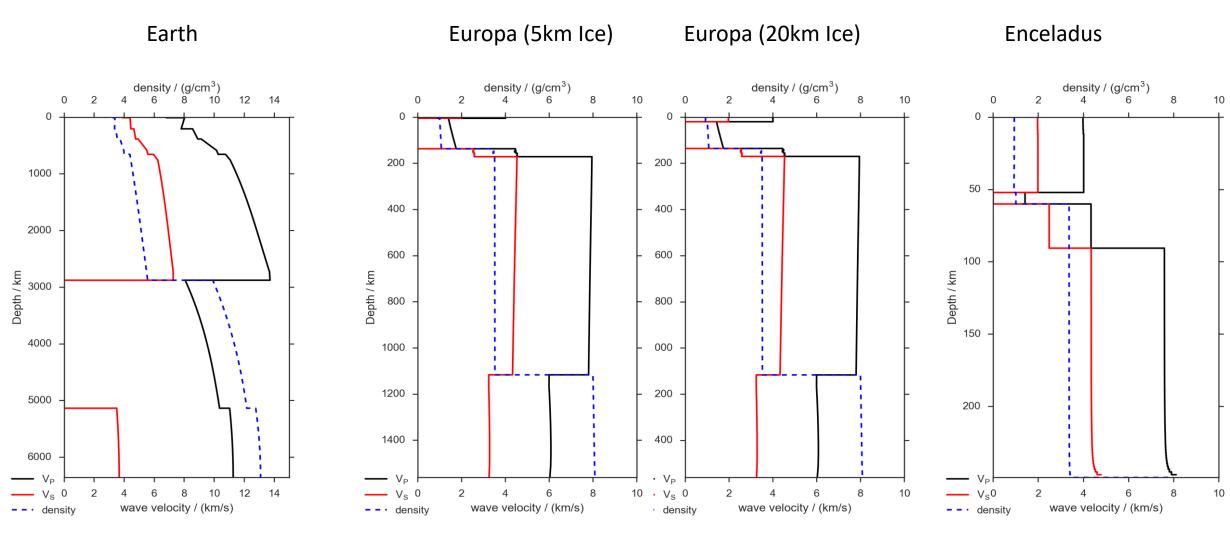


PlanetProfile

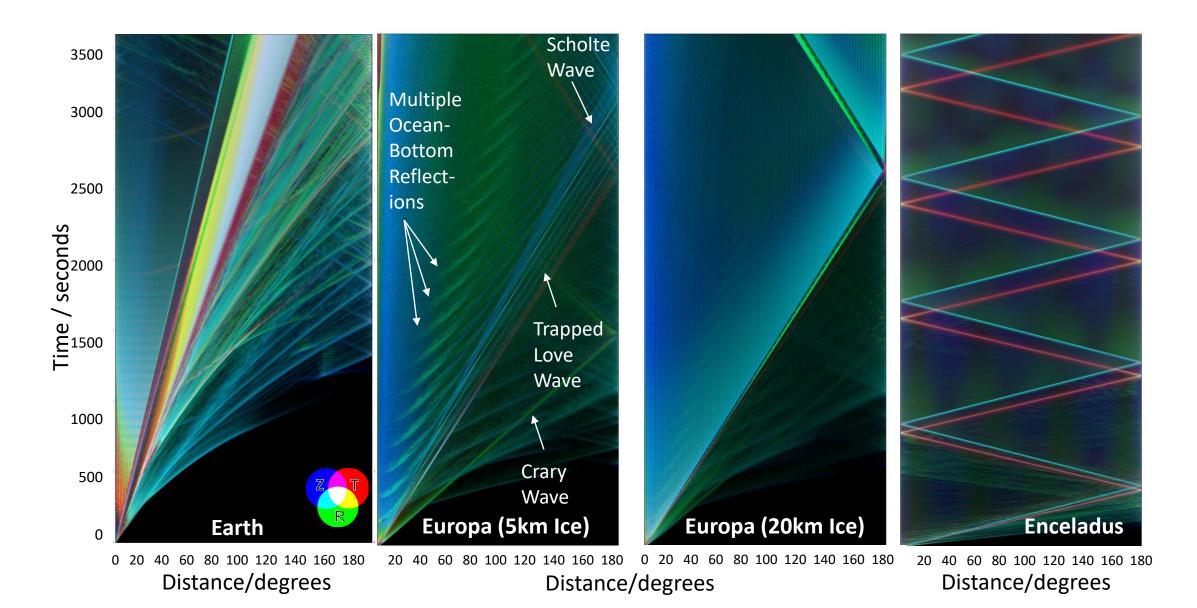
github.com/vancesteven/PlanetProfile



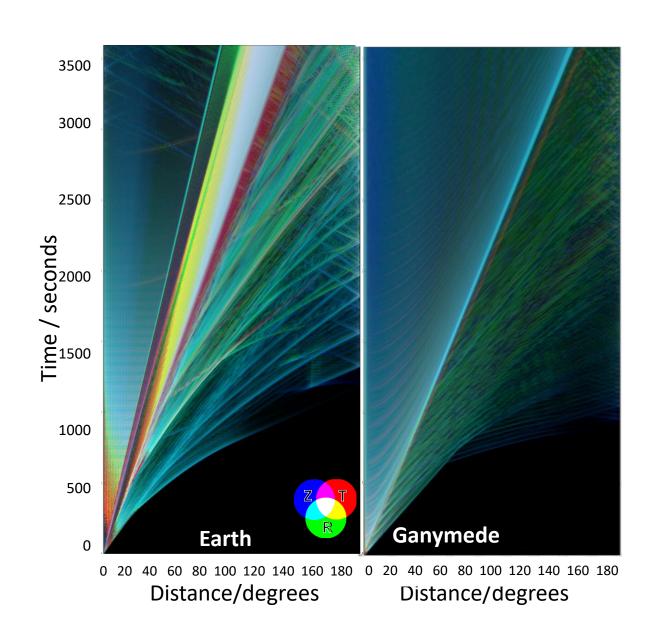
Seismic Models

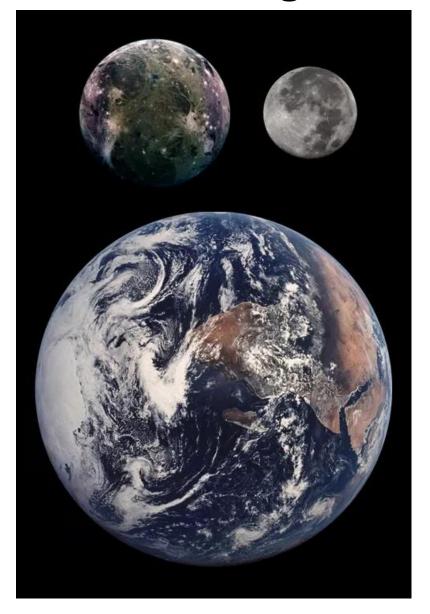


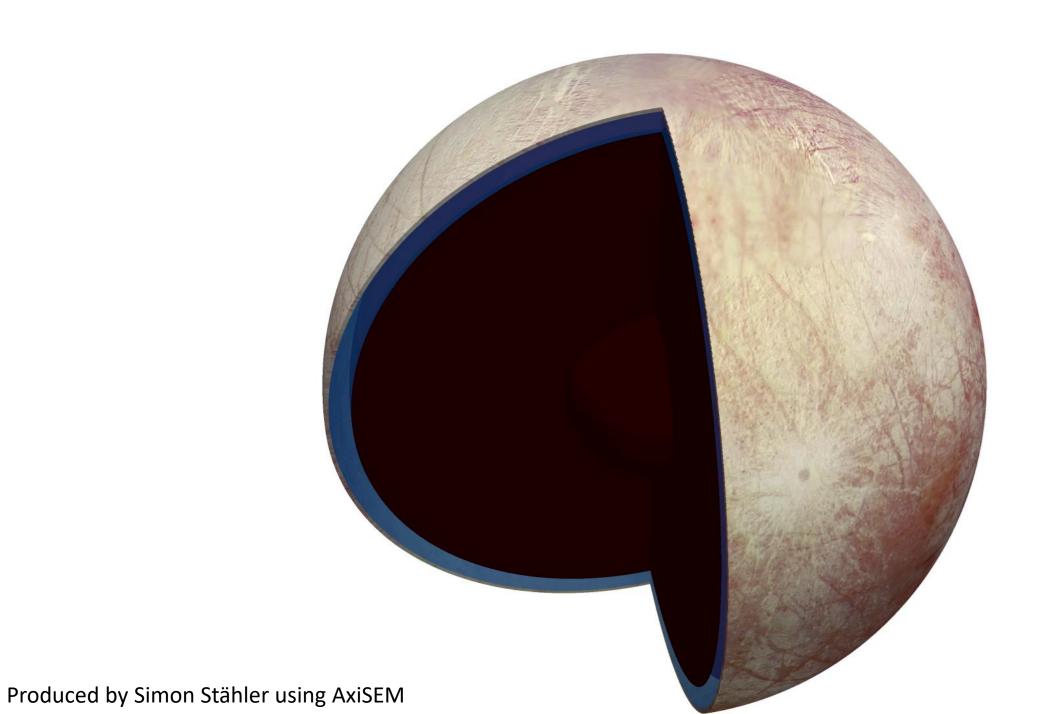
Each Ocean World Has Unique Seismic Signature



Each Ocean World Has a Unique Seismic Signature



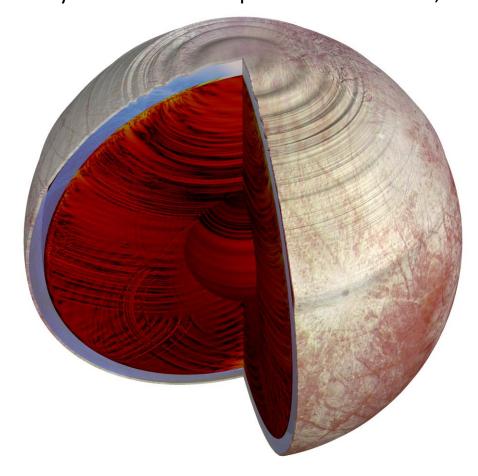




Estimating the Background Noise

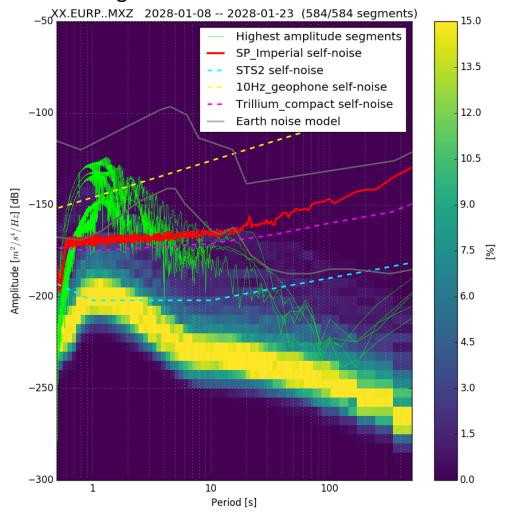
Panning et al., submitted

AxiSEM/Instaseis is used to compute a series of 1 week long noise records (figure 3), which are used to compute probabilistic power spectral density plots for different seismicity models and Europa structure models,



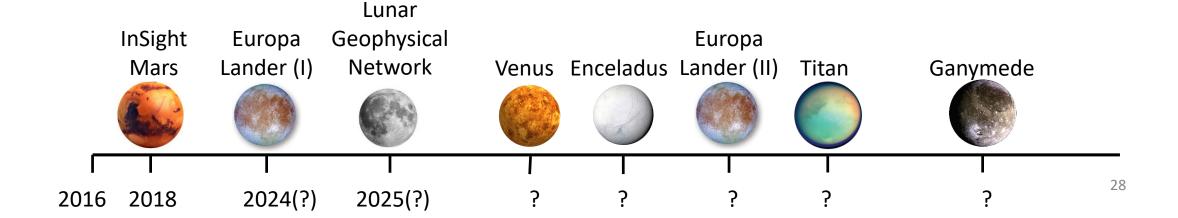
Assumptions:

- Tidal dissipation
- Largest moment release



Conclusions

- Icy ocean worlds are likely to be seismically active
- Seismic activity is likely to be varied, combining ice-quakes, tidal deformation, fluid induced seismicity (geysering, cryovolcanism, ocean circulation & acoustics)
- Each icy world has a unique seismic signature that can constrain habitability
- "Vital Signs" the unique signs of present day fluid motion may carry fundamental information about the depth and nature of volatiles + ocean circulation
- Future seismometers to ocean worlds must fit within tight power, mass and volume allotments of future landers, but be sensitive enough to maximize science return during exceptionally short missions



Overview of Seismic Sensing Technologies









Earth	Streckeisen STS-2	Trillium Compact	Geophone	MEMS Accelerometer*
Planetary	InSight VBB	InSight SP	Apollo Active Experiment	







~0.1Kg ~3cm



* Technology has not yet reached sensitivity requirements of Earth & planetary science

- ~0.01Kg ~1cm

- Very Broad Band (VBB)
- High Dynamic Range
- Ultra sensitive
- \$\$\$\$

- Broad Band
- High Dynamic Range
- Sensitive
- \$\$

- Narrow Band
- Limited dynamic range
- Strong motions
- \$

- Broad Band
- High dynamic range
- Sensitive
- 9

Backup

Planetary Seismology???

Prior Seismic Experiments

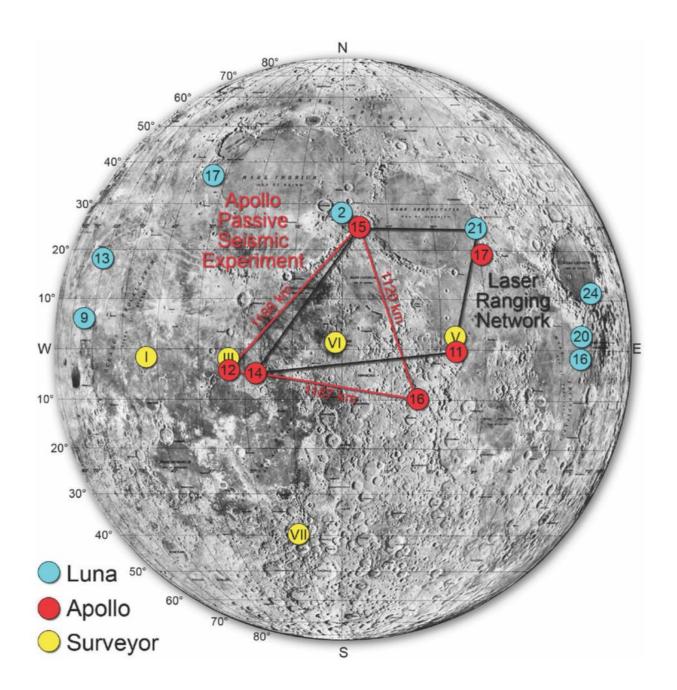
 Table 1
 Summary and history of planetary seismology experiments

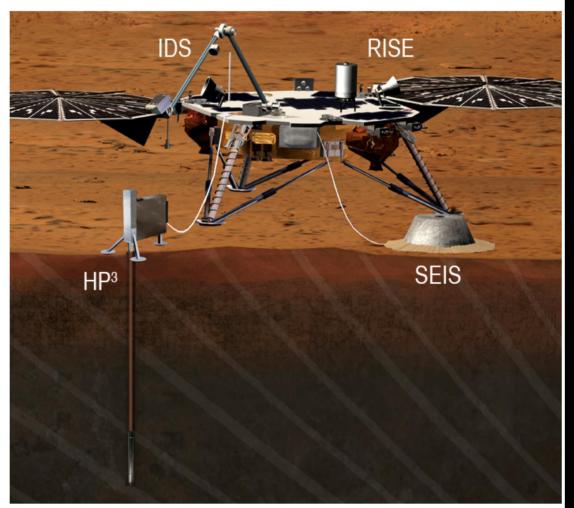
Mission	Launch	Major mission events	Instrument description	Seismometer deployment	Reference
Ranger 3	1962-01-26	Failure due to the booster. Moon missed	Vertical axis seismometer, with a free frequency of 1 Hz. (Mass: 3.36 kg)	Seismometer in a lunar capsule designed for a 130–160 km h ⁻¹	Lehner <i>et al.</i> (1962)
Ranger 4 1962-04-23		Failure of spacecraft central processor. Moon crash.		landing. Batteries powered for 30 days of operations	
Ranger 5	1962-10-18	Failure in the spacecraft power system. Moon missed.			
Surveyor	1966-1968	The seismometer was finally deselected from the payload of the Surveyor missions	Single short-period vertical axis seismometer (mass: 3.8 kg, power: 0.75 W)	Fixed to the lander.	Sutton and Steinbacher (1967).
Apollo 11	1969-7-16	Successful installation. Powered by solar panel, worked during the first lunation and stopped after 21 days	Passive seismic experiment (PSE). triaxis Long-Period seismometer (LP) and one vertical Short-Period	Installation performed by crew. Seismometers were manually leveled and oriented with bubble	Latham <i>et al.</i> (1969, 1970a, 1970b)
Apollo 12	1969-11-14	Successful installation of a network of 4	(SP) seismometer, with resonance	level and sun compass. A sun	
Apollo 14	1971-01-31	stations. For all but the Apollo 12 SP	periods of 15 s and 1 s, respectivel	protection/thermal shroud was covering the instruments. Power was delivered by a Plutonium thermal generator for A12-14-15-16	
Apollo 15	1971-07-26	seismometer and Apollo 14 vertical	(mass: 11.5 kg, power: 4.3-7.4 W)		
Apollo 16	1972-04-16	LP seismometer operated until the end of Sep 1977, when all were turned off after command from the Earth. 26.18 active station years of data collected.			
Apollo 13	1970-4-11	Moon landing aborted. No installation of the PSE experiment but lunar crash of the Apollo 13 Saturn-IV upper stage recorded by the A12 PSE.			
Apollo 14	1971-01-31	Successful installation and operation	String of three geophones on A-14 and	Geophones were anchored into the	Watkins and
Apollo 16	1972-04-16	of the active seismic experiments.	A16 and on four geophones on A-17.	surface by short spikes as they were	Kovach (1972)
Apollo 17	1972-12-07	Seismic sources were thumper devices containing 21 small explosive sources and a rocket grenade launcher with four sources exploding up to 1500 m on A-14 and A-16. Eight sources were used containing up to 2722 g of explosive and deployed at 3500 m by astronauts	Frequency was 3-250 Hz.	unreeled from the thumper/ geophone assembly.	Kovach and Watkins (1973a)

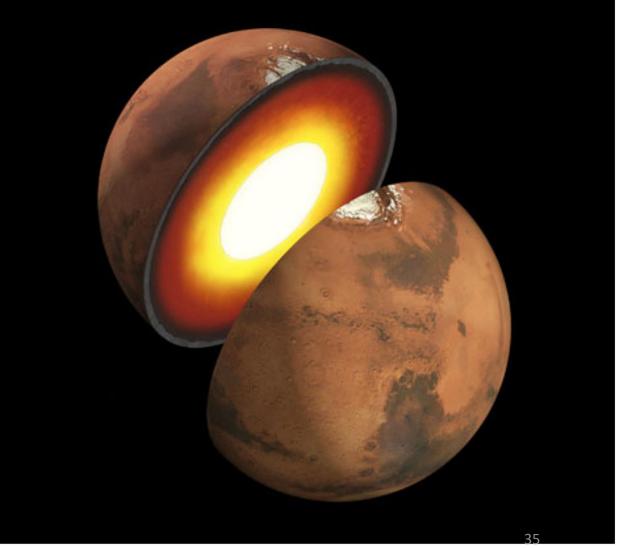
(Continued) 32

Mission	Launch	Major mission events	Instrument description	Seismometer deployment	Reference			
Apollo 17	1972-12-07	Deployment of the Lunar Surface Gravimeter. The gravimeter was unable to operate properly due to an error in the design of the proof mass.	Gravimeter designed for gravity waves detection. Additional long-period vertical seismic output (10 ⁻¹¹ lunar <i>g</i> resolution) for free oscillation detection, with a 16 Hz sampling.	Installation performed by crew.	Weber (1971)			
Viking Lander 1	1975-08-20	Successful landing but instrument failure.	Short-period instrument, with an undamped natural period of 0.25 s,	The seismometer was installed on the Lander platform. No recentering	Anderson et al. (1977a,			
Viking Lander 2	1975-09-09	Successful landing and 19 months of nearly continuous operation. Too high wind sensitivity associated to the elastic recovery of the <i>Viking</i> landing legs to the loading of the station by pressure fluctuations induced by the wind.	a mass of 2.2 kg, a size of $12 \times 15 \times 12$ cm and a nominal power consumption of 3.5 W.	was necessary since the three-axis seismometer had been designed to function even when tilted to up to 23°	1977b)			
Phobos 1-2	1988-07-07 1988-07-12	Respectively: Lost during transfer to Mars and Phobos; contact lost just before the final phase of lander deployment, after Mars orbit insertion		Instrument onboard the long-service lander.	Surkov (1990)			
Mars 96- Small surface stations	1996-11-16	Failure of the Block-D propulsion system in parking orbit. Earth reentry. Two small stations and two penetrators lost.	Long-period vertical axis seismometer (0.1–4 Hz, 0.405 kg for the sensor) combined to a magnetometer. 55 mW of power	Seismometer in the small surface station. Semi-hard landing (200g– 20 ms). Nominal operations of one Martian year with 90th first days of nearly continuous mode with internal batteries	Lognonné <i>et al.</i> (1998a)			
Mars 96 Penetrators			High-frequency seismometer (10– 100 Hz, 0.3 kg, 20 mW)	Seismometer in the penetrator. Hard landing. Nominal operations of one Martian year.	Kravroshkin and Tsyplakov (1996)			
Rosetta	2004-03-04	Landing on the comet 67P/ Churyumov-Gerasimenko planned a few months after rendezvous, expected on 22-05-2014	CASSE/SESAME experiment: High-frequency accelerometer covering the frequency bandwidth ~10-20 kHz.	Instrument mounted on the lander	(Kochan et al. 2000)			

Lognonné 2007







	Radius (km)	Density (kg m^{-3})	Moment of Inertia
Europa	1565.0 ± 8.0	2989 ± 46	$0.346 {\pm} 0.005$
Ganymede	$2631 {\pm} 1.7$	1942.0 ± 4.8	$0.3115 {\pm} 0.0028$
Callisto	2410.3 ± 1.5	1834.4 ± 3.4	0.3549 ± 0.0042
Enceladus	$252.1 {\pm} 0.2$	1609 ± 5	0.335
Titan	2574.73 ± 0.09	1879.8 ± 0.004	$0.3438 {\pm} 0.0005$

Fluid Properties

- MgSO₄ (aq) Vance and Brown 2013, Vance et al. 2014
 - 0-2 molal; 250-400 K; 0-2 GPa
- Seawater (NaCl) Gibbs Seawater (McDougall 2011)
 - 0-2S0; 250-400 K; 0-0.1 GPa
- NH3 (aq)
 - 0-10 wt%; 250-400 K; 0-2 GPa

Sound speed in ice

from fits to measurements around -35°C

Crystalline	K_S	K_S'	K_S''	μ	μ'	μ''
Phase	GPa		GPa^{-1}	GPa		GPa^{-1}
I_h	9.5	0.33	-0.026	3.3	0.537	025
II	13.89	1.6		5.15	3.5	
III	8.9	3.65		2.7	6.55	
V	11.8	4.8		5.7	0.9	
VI	14.6	4.1		5.0	3.0	

based on Shaw 1986 and Gagnon et al. 1988, 1990

$$V_S = \left[\frac{\mu}{\rho}\right]^{1/2}$$

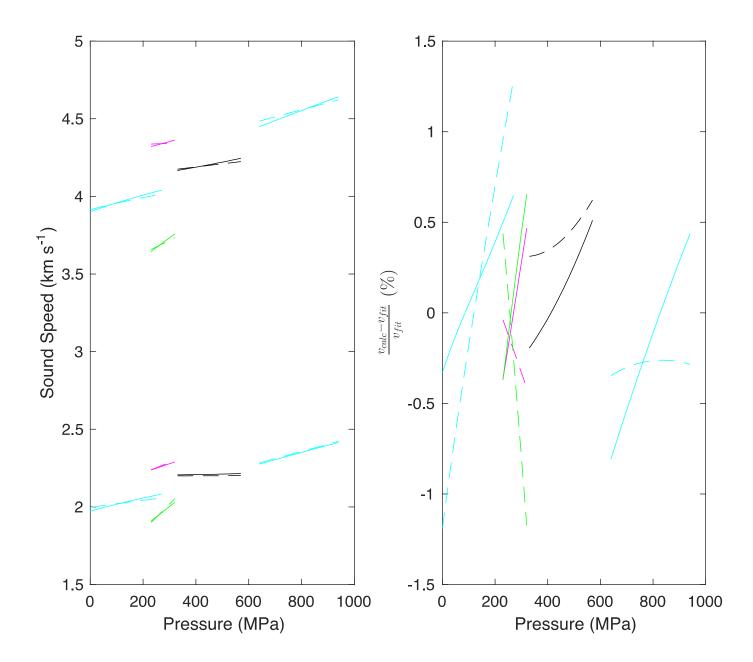
$$V_P = \left[\frac{K_S}{-} + \frac{4}{-}V_S^2\right]^{1/2}$$

forward model for bulk sound speed from ice equations of state based on density and phase boundaries

$$\frac{\partial \rho}{\partial P_T} = \frac{1}{v_c^2} + \frac{\alpha^2 T}{C_P}$$

Choukroun et al. 2010

Vance et al., in prep.



Earth 10² FBA-23 clip at 2g 10¹ 10° octave wide bandpassed acceleration Near Source ~ 10km Regional ~ 100km 10^{-3} 10² frequency - Hz

Instrument Performance Implication for Science Return

